







1 **Food production in space from CO₂ using microbial electrosynthesis**

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11
12 Abstract (248 words): Human space survival requires sustainable food production methods. The current food method
13 in space is launching prepackaged food which is costly, wasteful, and unsustainable. Alternatives include growing
14 crops and microalgae single cell protein (SCP) using artificial light photosynthesis, which are also costly and
15 inefficient. Prepackaged food and SCP food growing solutions were compared to microbial electrosynthesis of acetic
16 acid (MES-AA) using electroactive bacteria. The analysis employed an equivalent system mass (ESM) technique
17 customarily used by the National Aeronautics and Space Administration to compare and select from alternative
18 systems. Since the dominant cost of a space mission is the cost of launching mass, components of a system are
19 converted to an equivalent mass, including power, heat rejection, and volume. Distinct three-year roundtrip manned
20 missions were evaluated for the International Space Station, the Moon, and Mars. The average ESM of MES-AA is
21 1.38x and 2.84x lower than prepackaged food and microalgae SCP, respectively. The alternative food with the lowest
22 average ESM in space, SCP from hydrogen-oxidizing bacteria (HOB), is 1.86x lower than MES-AA. The expected
23 electricity-to-calorie energy conversion efficiency of MES is 19.8%; consuming 3.45 kW to fully feed five astronauts
24 with AA (diets would realistically compose multiple foods). MES-AA has a higher energy efficiency than any
25 currently investigated alternative food in space. The most promising food source is HOB, having the lowest ESM and
26 highest nutritional quality. However, MES can provide diet diversity at a lower cost than customarily storing
27 prepackaged food or growing crops in space.

28 Keywords (6): Equivalent system mass; Alternative food; Microbial electrosynthesis; Space food; Global catastrophic
29 risk; Existential risk

30 **1 Introduction**

31 The cost of providing food in space is excessively [1] high [2]. There is a need to develop a sustainable and
32 affordable food growing solution for environments in which it is difficult to grow food. Such environments include
33 the Earth's orbit, the Moon, or Mars, and even on the Earth in a sun-obscuring global catastrophe, such as a
34 supervolcanic eruption or a nuclear war causing cities to burn and the smoke inducing a nuclear winter. Existing space
35 food technologies are expensive, not scalable, and not suitable for long duration missions. This article investigates the
36 use of electroactive bacteria (EAB) as a resilient food production source for personnel in space. The goal was to
37 characterize the potential of a food production method from CO₂ and compare it to relevant technologies, in order to
38 mitigate certain restrictions of a manned space mission.

39 A number of challenges make agriculture in space a difficult premise, including lack of gravity, ionizing
40 radiation, limited on-board space, and contaminants [3]. There is a limit to greenhouse efficiency due to the
41 infrastructure required to protect from cosmic radiation. Current space foods are prepackaged food and fresh food
42 shipped to space after being grown on the Earth [4], neither of which are sustainable for long term space missions.
43 One alternative growing method to conventional agriculture is growing crops with artificial light, which is not
44 economically feasible for feeding a large population even on the Earth [5]. In addition, artificial light infrastructure
45 consumes large amounts of electrical energy, since phototrophic organisms (such as crop plants and microbes) have
46 low light to chemical energy conversion efficiency [6]. Since the dominant cost of a space mission is fuel [7],
47 minimizing the elements needed for food production in space is essential for reducing costs. Other potential solutions
48 not including plants or algae are foods from bacteria, which provide direct sources of nutrients needed by human
49 beings. EAB can release or retrieve electrons from solid state electron donors or acceptors using extracellular electron
50 transfer (EET) mechanisms. This allows them to use a wide range of electron donors/acceptors meaning that they can
51 survive in different environments [8]. There are several genera of EAB that grow under a variety of conditions on
52 Earth, from moderate to extreme environments, making them an ideal candidate for being grown in space. EAB are
53 also found naturally and widely spread across microbial communities, from wastewater systems to the human
54 gastrointestinal tract. Auto-lithotrophs, including some EAB, do not require organic molecules (such as sugar) or
55 valuable oxygen in order to grow. As far as we know, their environmental adaptability could give them added benefit
56 as a direct source of food if they are found to be edible like many other bacteria.

57 Bioelectrochemical systems (BES) are emergent technologies that convert organic and inorganic material
58 into electricity and/or chemical products through reduction and oxidation reactions [9]. BES can be divided based on
59 their applications; microbial fuel cells for the oxidation of organic/inorganic matter to the production of electricity,
60 and microbial electrosynthesis (MES) and microbial electrolysis cells, which reverse the microbial fuel cell process,
61 to product synthesis by applying an electric current [10]. BES results in a wide range of applications spanning from
62 energy production, waste recovery, bioremediation to food/nutrient production. The recent advancements of BES are
63 promising for assessing their use in manned space missions for the generation of electricity from human wastes and
64 the use of electricity for the synthesis of useful chemicals [11,12], moreover contributing to biologically sustained life
65 support. Such an adaptable design would allow the manned space missions to cut down the mass of shipped products

66 as they would be synthesized on board by MES. The use of MES in space for food and nutrient production has not
67 been explored yet. The advantages of a bioelectrochemical system in space are apparent by their efficiency when
68 compared to photosynthesis, the supply of electricity compared to energy from light, and the electrode surface area to
69 microbial cell surface area [13]. A study investigated the utilization of MES to produce food on the Earth to obviate
70 the use of additional stored food and artificial light photosynthesis during a global catastrophe [14]. While MES is
71 currently not a competitive food alternative for the Earth during a global catastrophe, the high caloric efficiencies of
72 this system may significantly reduce the equivalent system mass (ESM) for space food production. Thus, MES could
73 compete with conventional space food, prepackaged food, and be more energy efficient than a competing alternative
74 space food, microalgae SCP, specifically *Spirulina platensis*. *Spirulina platensis* [15] was selected to be grown in
75 space using artificial light photosynthesis because it is among the most productive photosynthetic microorganisms.
76 This study is similar to [1] which investigated the use of hydrogen-oxidizing bacteria (HOB) single cell protein (SCP)
77 as food in space. The current study assesses the economic feasibility of using MES with a mixed microbial community
78 in space for acetic acid (AA) production. AA is a product of fermentation; when diluted 4-6% by volume, is vinegar.
79 On the Earth, AA is commercially used in the manufacture of vitamins, antibiotics, hormones, organic chemicals, and
80 as a food additive [16].

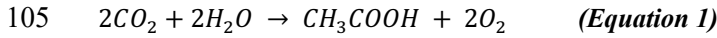
81 EAB have an advantage over other organisms because of their ability to utilize electrons from the electrode
82 surface on which they grow and transfer those electrons across their biofilm via extracellular electron transfer (EET).
83 A diagram of the MES process [14] is depicted in **Figure 1**, for which the overall chemical reaction is described by
84 **Equation 1**, based on partial reactions from literature [17]. Since fuel is the dominant cost of a space mission, and
85 fuel consumption is directly proportional to payload mass, the economic feasibility of MES in space was addressed
86 by using an ESM technique [18], established by the National Aeronautics and Space Administration (NASA).

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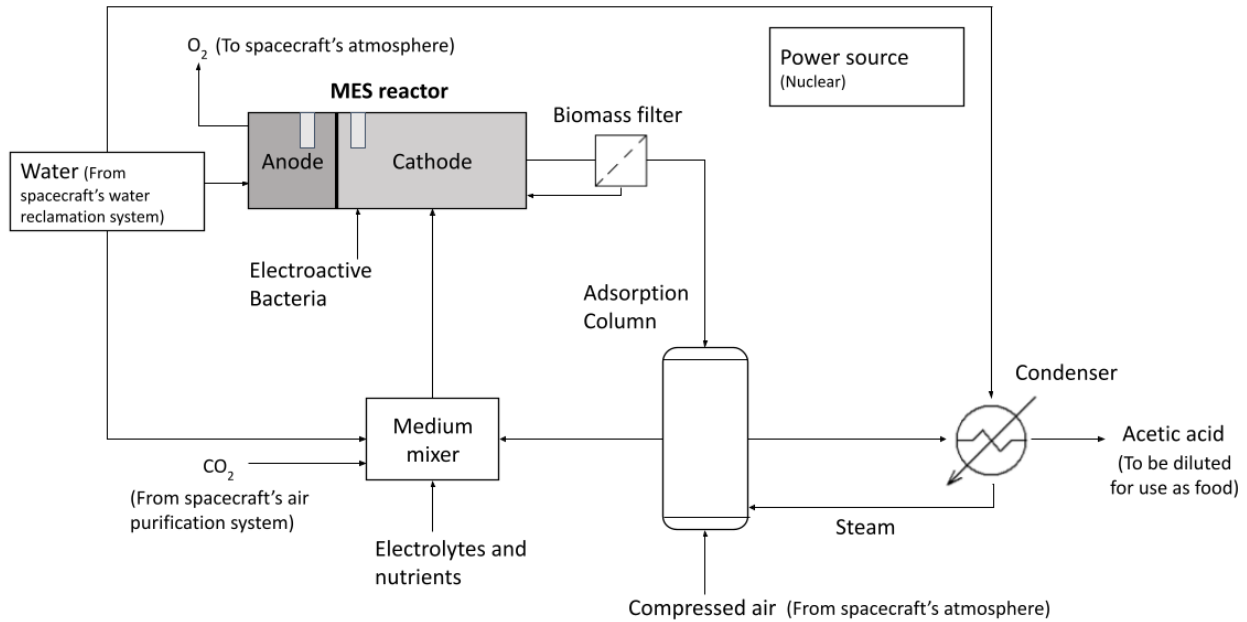
88 **2 Methods**

89 The ESM results were used to select between competing subsystems. In this study, the ESM of MES is
90 calculated and compared to the ESM of prepackaged food, microalgae SCP, and HOB SCP, calculated in a previous
91 study [1]. An International Space Station (ISS) mission, a Moon mission, and a Mars mission were assessed, each
92 consisting of 5 crew members (CM) for a duration of 3 years. Currently, more EAB research is needed to make
93 sufficiently accurate assumptions [19], but sufficient information exists to achieve the goals of this work, which is to
94 make a comparison to relevant space food technologies to inform prioritization of developments in the field. Values
95 need to be consistent with each other to evaluate the technology for the purpose of food production. MES-AA is treated
96 as the sole calorie source for this study to form an equitable comparison to other food production methods. In addition,
97 this food production method can be used as a resilient food for sun-blocking global catastrophic risk (GCR) scenarios
98 in accordance with its ability to use feedstocks not directly dependent on agriculture and sunlight, such as water and
99 industrial or atmospheric CO₂ [14]. Both CO₂ capture [20] and utilization [21] are rapidly expanding areas of research,
100 containing value in sustainable chemical energy production and energy integration. MES as a CO₂ utilization method

101 currently faces significant challenges such as low titers, high operating costs, and the achievement of industrially
 102 relevant production rates at an acceptable energy efficiency [22]. However, MES holds significant potential for
 103 production of value-added chemicals independent of fossil resources, which would contribute towards a circular
 104 carbon economy [19].



106



107

108 **Figure 1.** Process flow diagram of the MES-AA system adapted from [14]. The cathode compartment has a 3:1 volume
 109 compared to the anode.

110

111 2.1 MES system design and operation

112 MES technology has not yet been attempted in space applications for food/chemical production, however,
 113 EAB are known to survive in different environments using different substrates, which suggests potential suitability
 114 for microgravity environments. The design for the MES system is based on the most promising experimental lab-scale
 115 research ([23]). **Figure 1** illustrates a flow diagram of the MES food production concept which includes all major
 116 equipment, excluding storage tanks. Liquid reaction medium and CO₂ from the spacecraft's air purification system
 117 are mixed prior to their entry to the cathode and fixed at 32 °C. The cathode compartment includes the EAB, electrodes,
 118 and membranes needed for the MES reaction to occur. The reaction medium in both cathode and anode compartments
 119 contains trace elements and synthetic nutrient solution ([23]). The anolyte composition is optimized with decreased
 120 pH to favor proton crossing through the membrane. The inoculum used for EAB is considered to be a mixed microbial

121 consortium enriched with bacterial populations such as *Proteobacteria*, *Firmicutes*, *Bacteroidetes* and *Actinobacteria*
122 [24–26]. Experimental evidence suggests the use of aggregated forms of cells as a potential way to improve the
123 survivability of a microbial population in high-radiation microgravity environments and allow biofilms to thrive [27].
124 The EAB biofilm could be grown, optimized, and stabilized prior to the launch of the spacecraft to aid its adaptation
125 in space conditions. The reaction starts by applying a specific potential, -0.85 V vs standard hydrogen electrode (SHE),
126 to achieve AA production. The liquid mixture is then moved to an EAB separator (biomass filter) where any remaining
127 EAB are filtered and collected to be recycled to the cathode compartment, followed by product separation through an
128 absorption column. Any byproduct production such as O₂ and H₂O can be recycled within the MES system or used
129 for the needs of the spacecraft.

130

131 **2.2 Equivalent system mass**

132 ESM equates all components of a subsystem with a unit mass for analog comparison using mass
133 equivalencies, derived by the NASA Life Support Baseline Values and Assumptions Document (BVAD) [28]. The
134 relevant equation for this study, originally from the BVAD [18] and adapted [1] to fit a time-independent study, is
135 indicated by **Equation 2**. From **Equation 2**, initial (apparent) mass M_I comprises the initially launched mass and
136 ancillary components required for food production; power consumption P is the electrical power demand of each
137 system; cooling (or heat rejection) C is the thermal demand for rejecting heat developed by the system (since it is a
138 closed system, $C = P$); and initial (on-board) volume requirement V_I is the volume occupied by the food system at
139 launch. Mass equivalency factors P_{eq} , C_{eq} , and V_{eq} convert non-mass components to mass equivalencies and are
140 determined by the design and cost of infrastructure. For example, the mass equivalency factor for power P_{eq} relates
141 the quantity of mass and cost of installation and components per rate of power delivered, so that the product of P
142 (kW_{electrical}) and P_{eq} (kg/kW_{electrical}) is unit mass (kg). The location factor L_{eq} is dependent on the spacecraft
143 infrastructure and the destination in reference to the acceleration needed to travel from the Earth's surface to Low
144 Earth Orbit (LEO) [28].

$$145 \quad ESM = L_{eq} * [(M_I) + (P * P_{eq}) + (C * C_{eq}) + (V_I * V_{eq})] \quad (\text{Equation 2})$$

146

147 **2.2.1 Apparent mass and volume**

148 The apparent mass of the MES-AA food production system includes the mass of the reactor medium and the
149 equipment. The MES equipment includes a reactor, separation equipment (adsorption column, sorbent, water boiler),
150 support equipment (pumps, pipes, frames, control equipment), and the power source. The medium mass is the largest
151 constituent of the setup mass and depends on the volumetric productivity of the system. The MES key performance
152 parameters used in this study to estimate system mass and energy usage are based on the system characterized by the
153 most promising lab-scale research [23]. Although other studies have shown somewhat higher production rates, the

154 reference study has better overall performance provided by the maximum production rate ($11.55 \pm 0.15 \text{ kg/m}^3$
155 catholyte/d), product titer ($8.2 \pm 0.4 \text{ kg/m}^3$ catholyte), and electron recovery ($91 \pm 9\%$) [23]. However, the AA
156 production rates that have been sustained in prolonged operation over hundreds of days are lower than the maximum
157 at around $5\text{-}9 \text{ kg/m}^3$ catholyte/d [23], so this study conservatively uses this range as a basis. The associated catholyte
158 volume ($\text{m}^3_{\text{catholyte}}$), which is the portion of electrolyte solution in the cathode compartment, is calculated as the ratio
159 of production rate requirement (kg/d) and volumetric productivity ($\text{kg/m}^3_{\text{medium/d}}$). **Figure 1** indicates that 75% of the
160 volume is in the cathode [23]; the reactor volume is calculated by dividing the associated catholyte volume by this
161 percentage. The corresponding values of volumetric productivity in terms of medium volume are then $(3.75\text{-}6.75$
162 $\text{kg/m}^3_{\text{medium/d}})$. The total setup mass was estimated as the average value obtained from the low end and high end of the
163 volumetric productivity range, as described in the Supplementary Material. The reactor volume is estimated using
164 volumetric productivity and required mass flow rate, then multiplied by 1.5x to account for the volume occupancy of
165 the units around the reactor and ancillary equipment. The total system volume is estimated to be 1.26 m^3 .

166

167 2.2.2 Caloric energy conversion and electrical power requirement

168 A nuclear reactor was the selected power source for this study considering it has less equivalent mass than a
169 solar powered system for these particular missions and produces steady power [1]. The electrical power demand
170 depends on the electrical-to-caloric energy efficiency of the system, or in other words, the electricity input required to
171 obtain a calorie in food. The electrical power requirement of the MES setup is calculated using **Equation 3**. From
172 **Equation 3**, \dot{m}_{AA} (kg/d AA) is AA mass flow rate required to feed 5 CM exclusively with MES-AA, V (Volts) is
173 voltage, F is the Faraday constant ($96,485 \text{ coulombs/mol}$), n_{e^-} is number of electrons required per unit of AA produced,
174 FE is Faradaic efficiency, and X_{AA} is the fraction of acetate in the product distribution. Contributors to energy
175 consumption of the system were considered: (1) energy requirement of MES, or the energy required to make the
176 bacteria produce the required amount of AA, (2) energy requirement of separation, in this case to separate AA from
177 the reactor medium, and (3) energy requirement of CO_2 capture needed as starting material for AA production. All
178 values applied to calculate energy consumption are compiled in **Table 1**. Values were gathered from experimental
179 literature [29], considered as representative of current state of science MES systems. The proposed AA recovery
180 system via adsorption requires energy mainly to produce the steam used to desorb the AA from the sorbent, with an
181 expected requirement of 5 units of steam per unit of AA recovered [30]. A typical water boiler efficiency of 80% is
182 used. Regarding CO_2 capture, the system at the ISS is used as a reference with 20% thermodynamic efficiency [31].
183 An overall CO_2 utilization of 95% was used, close to single pass conversion values reported in literature [32,33],
184 which could be increased even further from recycling unconverted CO_2 .

$$185 \quad P = V \cdot F \cdot n_{e^-} / (FE \cdot X_{AA}) \cdot \dot{m}_{AA} \quad (\text{Equation 3})$$

186

187

Variable	Value	Unit	Basis for
Faradaic efficiency into products	70%		MES-AA from CO ₂
Cell voltage	3	V	MES-AA from CO ₂
Share of acetate in product distribution	100%		MES-AA from CO ₂
Electrons required	8	mol e-/mol AA	MES-AA from CO ₂
Steam requirement of desorption	0.2	kg AA/kg steam	Separation of AA from reactor medium
Latent heat of water	2,256	kJ/kg steam	Separation of AA from reactor medium
Boiler efficiency	80%		Separation of AA from reactor medium
Overall CO ₂ utilization	95%		CO ₂ capture
ISS CO ₂ capture system efficiency	20%		CO ₂ capture
Minimum CO ₂ capture energy requirement	21	kJ/mol CO ₂	CO ₂ capture

189

190 **2.2.3 Location factors**

191 A location factor is needed for each acceleration during a mission to account for time-dependent changes in
 192 mass. Since this study is independent of time, one location factor was derived for each mission using values from the
 193 BVAD [28]. There are six acceleration steps that are relevant to this study, including (1) Earth’s surface to LEO, (2)
 194 LEO to the mission body’s orbit, (3) mission orbit to mission surface, (4) mission surface back to mission orbit after
 195 completing the mission objective, (5) mission orbit to LEO, and (6) LEO to the Earth’s surface. The ISS mission
 196 would only involve the reference location factor of the Earth’s surface to Low Earth Orbit (LEO), where $L_{eq,(1)} = 1.0$.

197

198 **3 Results**

199 Volumetric productivity of the MES system corresponds to a volume requirement of 0.6-1.1 m³_{medium} for the
 200 desired AA production capacity. The corresponding medium mass would be 0.6-1.1 tonnes, with the accompanying
 201 equipment mass estimated to be much lower in comparison (**Table 2**), and mass of the nuclear reactor equaling 79 kg
 202 from an infrastructure penalty of 23 kg/kW_{electrical} [28]. Feeding 5 CM for 3 years would require 15.5 million kcal.

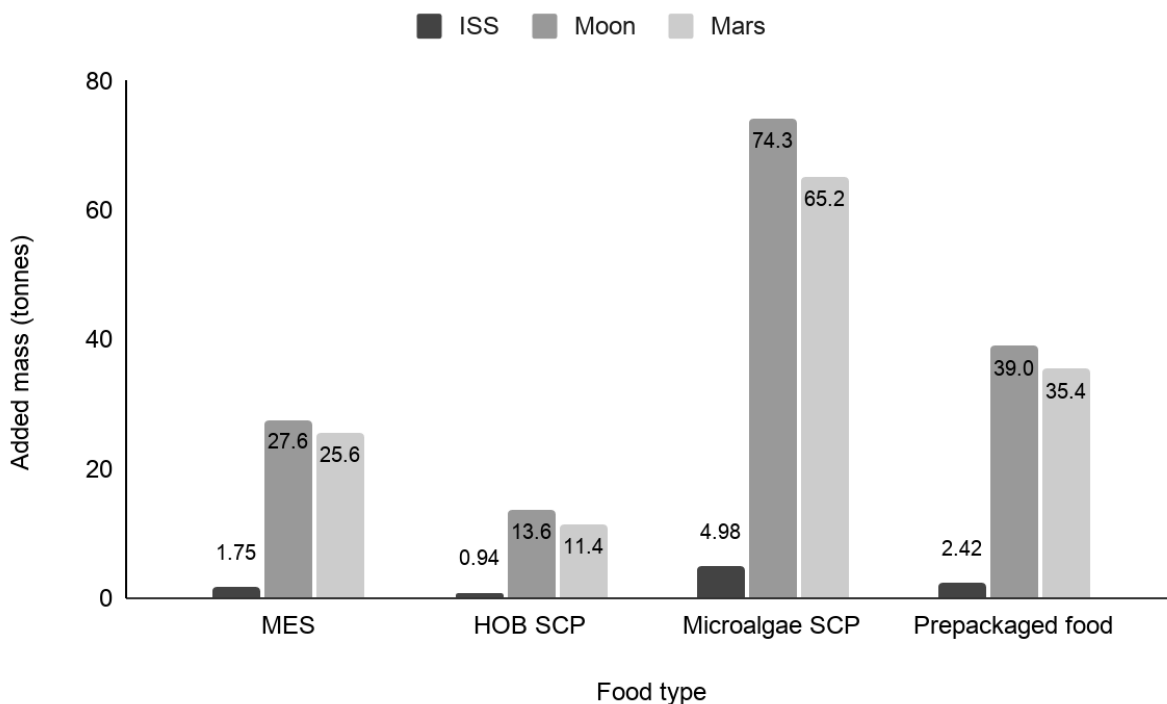
203 Since AA energy density is 3.49 kcal/g [34] and astronaut food requirement is 2,820 kcal/d/CM [28], then the
 204 equivalent mass flow rate \dot{m}_{AA} would be ~ 4 kg/d of AA production. The estimated power requirement to produce the
 205 AA calorie-equivalent of feeding 5 CM is 3.45 kW, with an overall electricity-to-calorie efficiency of 19.8%. Nearly
 206 80% of the power demand is required by AA electrosynthesis.

207

208 **Table 2.** Average expected MES setup mass.

System part	Mass (kg)
Nuclear Reactor	79
Reactor Medium	839
Empty reactor mass	60
Adsorption column + adsorbent mass	10
Boiler mass	10
Support equipment (pipes, pumps, frames, instrumentation, etc.)	80
Total setup mass	1,078

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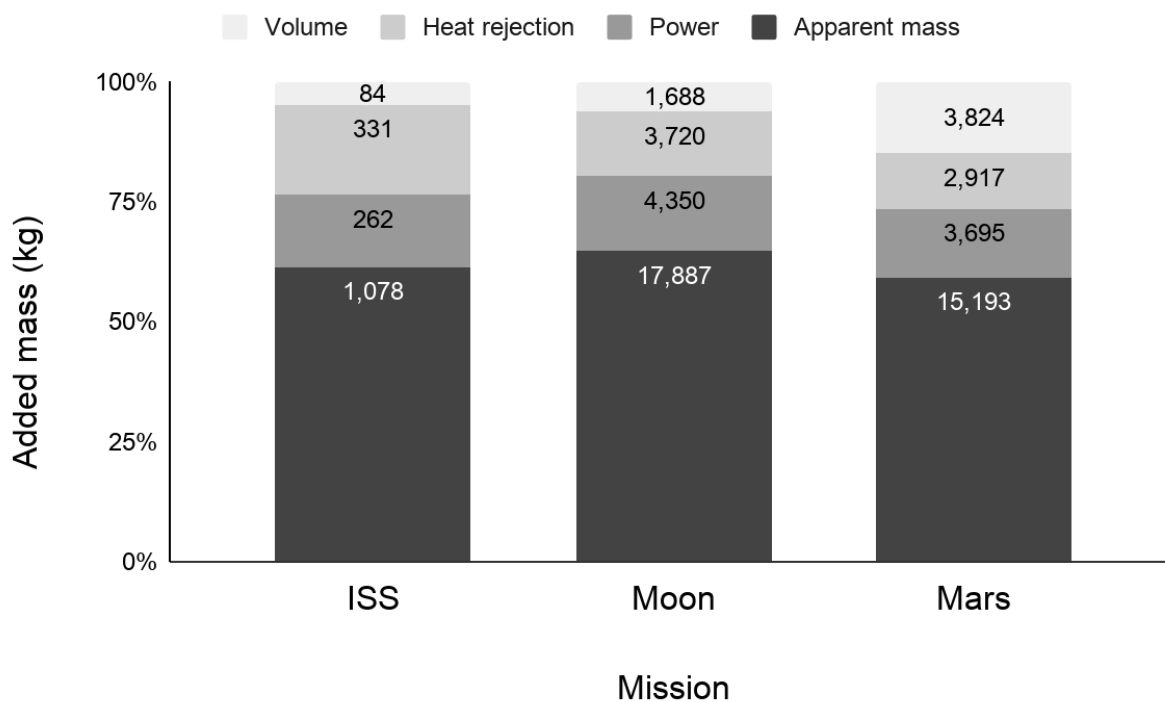
211 **Figure 2.** The ESM of the alternative space food systems for each mission. Values are represented as added mass (in
 212 tonnes) to the mission.

213

214 **Table 3.** Components of each food alternative (ordered from largest ESM food to smallest), in kilograms, after
 215 applying mass equivalency factors to non-mass units and before applying location factor (which acts as a mission
 216 scalar). This may be used for comparison within each mission scenario, not for comparison between missions.

	Apparent mass	Power	Heat rejection	Volume
Microalgae SCP				
ISS	1,918	1,300	1,642	119
Moon	1,918	1,300	1,112	144
Mars	1,918	1,300	1,026	384
Prepackaged food				
ISS	1,842	215	272	88
Moon	1,842	215	184	107
Mars	1,842	215	170	284
MES				
ISS	1,078	262	331	84
Moon	1,078	262	224	102
Mars	1,078	262	207	271
HOB SCP				
ISS	273	293	371	4
Moon	273	293	251	5
Mars	273	293	232	13

217



218

219 **Figure 3.** Component contributions to the ESM of MES for each mission before summing them as a mission ESM and
 220 after applying mission location factors and mass equivalency factors to power, heat rejection, and volume. Values on
 221 bars are in kilograms (kg).

222

223 4 Discussion

224 The methodology still requires further development and testing in space conditions before implementing.
 225 Improvements to EAB studies and related fields are currently advancing [35–37]. Before this production method is
 226 sufficiently advanced for the proposed application, there are several technical challenges relating to its reliability as a
 227 food production method. The appropriate cathode material should be identified to enhance the electrocatalytic activity
 228 and microbial cell adherence to the cathode. The appropriate bacterial strain or community should be identified
 229 specific to its food production employment, i.e. for AA production; perhaps genetically modified to improve
 230 production rates. Condition monitoring should be performed to optimize energy consumption, biofilm formation time,
 231 biofilm stability, and byproduct production rates. Studies performed on microorganisms in microgravity confirmed
 232 the increased production of viable cells, biofilm biomass, and thickness compared to normal gravity [38]. These
 233 promising results are an advantage to BES since a healthier biofilm formation will drive better EET and therefore
 234 higher efficiency rates (synthesis). Additionally, the effects of spaceflight on the physiology of EAB, specifically
 235 *Shewanella oneidensis* manganese-reducing strain (MR-1), have been examined during transfer to the ISS [11].
 236 Potential impacts of a space environment have been thought to be an increase in cellular stress, which could negatively

237 impact growth performance, and a decrease in biofilm development, which could negatively impact growth and
238 byproduct formation. However, studies including electroactive cultures must be performed under space conditions to
239 verify this behavior. Field experiments in microgravity conditions with an MES food production system are required
240 to establish feasibility for manned space missions.

241

242 **4.1 Alternatives comparison**

243 This analysis indicates that the ESM of MES is lower than prepackaged food and microalgae (**Figure 3**). The
244 ESM of HOB remains the lowest alternative, and therefore lower cost compared to each food system. The mean ESM
245 of MES is 2.84x lower than microalgae, 1.38x lower than prepackaged food, and 1.86x higher than HOB SCP. The
246 apparent mass is the dominant contribution to the total ESM of MES for each mission; whereas on-board volume is
247 relatively insignificant for ISS and Moon missions (**Figure 4**). ESM results are higher in value for the Moon mission
248 than for Mars (**Figures 3-4**) due to location factors. This is because the Moon and Mars missions have different
249 shuttles, propulsion types, and transportation history (i.e. whether payloads are jettisoned during travel). ESM is not
250 the exclusive metric for a comparison study since it lacks considering reliability, safety, and palatability [39].
251 However, it is pivotal for evaluating cost. The location factor is multiplied as the endcap of the analysis to magnify
252 the propulsion requirement. As MES productivity advances, MES-AA would become more competitive as a food
253 source in space. As the technology of MES advances and volumetric productivity and faradaic efficiency are
254 improved, the apparent mass of an MES setup will reduce and could come closer to (or optimistically pass lower than)
255 the ESM of HOB. Based on current, long-term sustained literature values for MES (volumetric productivity of 6.75
256 $\text{kg}/\text{m}^3_{\text{medium}}/\text{d}$ and faradaic efficiency of 90%) the ESM of HOB is more favorable than for MES. In order to make
257 MES competitive with HOB in terms of ESM (for the expected faradaic efficiency of 70%), volumetric productivity
258 equal to 25 $\text{kg}/\text{m}^3_{\text{medium}}/\text{d}$ or higher would be needed. High volumetric productivities have been achieved [40];
259 however, they were not used in this analysis because it is not the exclusive key parameter; this system had much lower
260 titers and overall lower energy efficiency. Another advantage to the system used as a basis in this study is that it was
261 tested in long-term operation (hundreds of days). The electricity-to-calorie efficiency for MES-AA was calculated as
262 19.8% which is more favorable than the two competing microorganisms. The same conversion is 4.4% efficient for
263 microalgae SCP and 17.7% efficient for HOB SCP [1]. It is worth mentioning that the range of 50-90% faradaic
264 efficiency translates to 15.2-20.8% overall electricity-to-calorie efficiency, but the average expected value of 70%
265 faradaic efficiency corresponds to the 19.8% electricity-to-calorie efficiency value.

266

267 **4.2 Life support considerations**

268 Food is a category of life support and can be supplied via MES using resource recycling. Life support
269 subsystems are currently in place to improve air and water recovery and waste recycling but have low effectiveness
270 [41]. **Table 4** compares each component of MES to prepackaged food and microalgae; of which the microorganisms

271 display higher effectiveness with resource recovery and recycling. Ammonia, providing the nitrogen for EAB, can be
272 recycled from urine. Water, consumed in electrolysis and microbial growth, can be supplied by the water production
273 in astronauts' metabolism (mostly urine, perspiration, and exhalation); oxygen as a product of electrolysis can be used
274 for astronauts to breathe. The carbon in CO₂ produced by the astronauts' metabolism is consumed by microbial growth
275 and MES. Net MES activity (food production) can be stable with complete carbon recycling; meaning less expected
276 amounts of raw materials would be needed at the beginning of a mission leaving the Earth. These would include
277 specific nutrients for the bacterial culture, and the amounts required are small enough to be negligible in terms of mass
278 at launch. Resources such as CO₂ and water may also be available from mission locations, particularly on Mars. The
279 prepackaged food option additionally requires operation of a Sabatier reactor that combines CO₂ produced by
280 astronauts with H₂ to yield methane waste and water; oxygen is recycled via electrolysis of the water product [42].
281 The methane and excess water wastes add mass to the system and must be ejected. Although MES contributes to
282 biologically sustained life support, the pressure and atmospheric conditions would still need to be controlled. In
283 comparison with the current prepackaged food standard, the MES-based life support system substitutes for the
284 Environmental Control and Life Support System (ECLSS) equipment, but both systems would incorporate an identical
285 set of equipment (Sabatier, etc) as a backup. Thus, the backup does not need to be accounted for in either option, but
286 the ECLSS system needs to be accounted for as an ESM contribution of the prepackaged food option. Frozen EAB
287 starter cultures may be carried in the spacecraft for increased resilience of the MES life support system, but this was
288 not accounted for in terms of mass.

289

290

Component	MES	Prepackaged food	Microalgae	HOB
H ₂ O (liquid)	Consumed in the MES process	Consumed in drinking, electrolysis, and rehydrating food, and product of crew metabolism	Consumed by microbial growth	Consumed in electrolysis and microbial growth
CO ₂	Consumed for microorganism growth and food production	Product of crew metabolism, converted to methane waste via the Sabatier system and ejected	Consumed for growing microorganisms for food	Consumed for growing microorganisms for food
O ₂	Product of MES	Produced via electrolysis	Product of microbial growth	Product of electrolysis
Food	Vinegar	Prepackaged food	<i>Spirulina platensis</i> powder	Microbial protein
Additional infrastructure	MES setup, power supply	ECLSS	Microalgae setup, power supply	HOB setup, power supply
Additional power requirements	Bioreactor, separation, processing of AA	ECLSS	Bioreactor, drying, processing of SCP	Bioreactor
Additional thermal control	Bioreactor cooling, heat exchangers, power supply cooling	ECLSS	Bioreactor cooling, heat exchangers, power supply cooling	Bioreactor cooling, heat exchangers, power supply cooling
Crew time	Operating, maintenance, cleaning (neglected)	Maintenance, cleaning (neglected)	Operating, maintenance, cleaning (neglected)	Operating, maintenance, cleaning (neglected)
Waste	Non recyclable waste from spent media (if any)	Food packaging, human waste and contaminants, methane gas, excess water	Non recyclable waste from spent media (if any)	Non recyclable waste from spent media (if any)

292

293 **4.3 Nutrient diversity and future work**

294 This study explored MES-AA as the exclusive calorie source. A proper nutrition plan may consist of
295 significant portions of MES-AA but since this source does not provide all of the nutritional needs of a human, it would
296 need to be complimented by food sources that provide vitamins, minerals, fats, carbohydrates, and proteins. These
297 needs may be met in combination with other alternative foods such as prepackaged food, microalgae, HOB SCP [1]
298 or non-biologically synthesized carbohydrates from CO₂ (García Martínez et al. to be published). The latter is the last
299 work on the current series of publications on alternative foods for space missions and contains a more holistic
300 comparison of the five food options mentioned above. Further research can be derived from this, moreover, combining
301 low-cost alternative foods for a mission. MES has been studied as an alternative food production method to feed
302 humanity during a global catastrophe inhibiting conventional agriculture, constituting a global catastrophic risk [14].
303 Such a catastrophe could potentially provoke or lead to existential risk, threatening the survival or future potential of
304 humanity.

305 Alternative foods inherently produce food at lower relative cost or energy consumption than the most
306 apparent solutions: stored food and crops grown with artificial light photosynthesis. This is analogous to prepackaged
307 food and artificial light-grown microalgae. Refuges have been considered for reducing existential risk to humanity by
308 increasing the likelihood of recovery from a global catastrophe or for self-sufficiency [43]. The small scale MES food
309 solution may relate to a refuge or small community on Earth during a sun-inhibiting global catastrophe, similar to a
310 refuge in space. Furthermore, this may apply to off-grid communities, such as those in rural Alaska [44], in which the
311 cost of energy and product delivery is very high (analogous to current prepackaged food delivery to the ISS).

312 An additional option for food production from BES could be consumption of the EAB as a form of SCP,
313 similar to the HOB SCP produced from H₂ gas described in the first article of this series [1]. This option was ultimately
314 not developed here since producing SCP from EAB as a space food would likely be worse in terms of ESM than SCP
315 from HOB. It is unlikely that a different electron donor (namely cathode electricity rather than H₂) would increase the
316 metabolic rate of CO₂ fixation. The reason is that the same metabolic pathways are used once the electrons get into
317 the cells [45], which means the cell growth rate will almost certainly be lower than that of HOB using external H₂.
318 This would result in a lower production rate and thus higher ESM.

319 Researching the cultivation of EAB for SCP production and its composition could be useful on the grounds
320 of nutrient diversity. Even though EAB SCP has higher ESM than HOB SCP, it could still be useful in combination
321 if it could provide useful nutrients not found in the HOB SCP. However, no nutritional analysis of EAB SCP can be
322 found currently in literature. Bacteria in general have high protein content (30–60% of dry weight) and some have
323 high lipid content (50-70% of total biomass dry weight) [46]. In addition, different types of bacteria contain vastly
324 different micronutrient profiles, potentially including essential fatty acids [46], while generally having a high average

325 caloric content (over 5 kcal/g dry) [47]. Producing a variety of microbial foods could contribute to efficiently
326 achieving a balanced diet in space missions.

327 Even though the use of MES seems promising, further research needs to be performed to ensure the ability
328 and efficiency for food production in space. The survivability of EAB in space will directly affect the operation of the
329 BES and therefore the production rates. To the best of our knowledge, no studies have been performed to test the long-
330 term survivability of EAB in space during BES operation, with the longest one lasting only 4 days [11]. However,
331 results from non-EAB studies showed that certain bacteria not only can endure extreme environments and survive in
332 space conditions but also demonstrated improved growth and activity rates [27,48]. *Pseudomonas aeruginosa* has
333 shown that spaceflight promotes biofilm formation and increases the numbers of viable cells, biomass, and thickness
334 of biofilms [38]. Although these studies showed promising results, cellular stress, fluid, proton exchange membrane
335 longevity, and gas distribution and separation must be studied prior to the application of BES in space [49].
336 Furthermore, a food production reactor based on MES is yet to be developed but nevertheless space food production
337 is an additional potential application for this technology. Laboratory scale experiments consisting of a targeted species
338 of EAB have been difficult to manage [50]; however, a mixed culture of acetogenic EAB is more robust to
339 contamination by non-electrochemical microorganisms. Although the MES method discussed here is potentially
340 viable, this study is based on a theoretical process that has yet to be performed at this scale [29]; therefore, key
341 performance values may vary.

342

343 **5 Conclusions**

344 This analysis determined that MES-AA is a cost-reducing food substitution in space compared to
345 conventional prepackaged food and using artificial light photosynthesis to grow microalgae, which is among the most
346 productive photosynthetic organisms. Food produced via MES-AA would be at least one order of magnitude less cost
347 than growing conventional crops in space with artificial light, and significantly more energy efficient. The conversion
348 efficiency of electricity to caloric energy for MES-AA was calculated to be 19.8%. Fulfilling the caloric needs of a 5-
349 crew mission, this would draw 3.45 kW from the system, all of which would eventually need to be rejected as heat.
350 In view of this favorable conversion efficiency, the electrical and thermal demands on MES-AA is lower than every
351 alternative food system considered in this study. However, the apparent mass is still relatively high due to low values
352 of MES volumetric productivity, which could potentially be improved via further research in MES for AA production.
353 For the latter reason, the ESM of HOB SCP, which leads the competition of food in space, is 1.86x lower than MES.
354 However, the final ESM of MES is on average 1.42x lower than prepackaged food and 2.84x lower than microalgae
355 SCP, and therefore the cost is proportionally lower. Although MES in space (and on the Earth) has significantly lower
356 cost and energy consumption than these alternatives, it must be complemented with other foods. This MES food
357 solution contributes to a biologically sustainable life support system by recycling CO₂ into useful (edible) compounds
358 and recycling human contaminants, conducive to closed systems for manned space missions.

360 **Funding**

361 This research was supported in part by the Alaska Space Grant Program (ASGP) at the University of Alaska Fairbanks
362 (from the National Aeronautics and Space Administration - NASA) (Cooperative Agreement 80NSSC20M0070) and
363 the Alliance to Feed the Earth in Disasters (ALLFED). ASGP was involved in study design and ALLFED was involved
364 in all parts of this research.

365

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